

Performance Evaluation of Surface Riding Wave Energy Converter with Linear Electric Generator

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Abstract

In this study, we devised a new WEC (wave energy converter) called SR-WEC (Surface Riding WEC). The SR-WEC consists of two bodies: the outer cylinder with an armature assembly (body #1) and a magnet assembly (body #2) sliding inside the armature. For the SR-WEC, the relative sliding displacement and velocity are caused by gravity acceleration and the outer cylinder's motions, and they lead to electrical power generation. To evaluate its performance, a numerical simulation tool was developed, which solves the fully-coupled floater-mooring-generator dynamics. During the developing stage, the appropriate hydrodynamics model, sliding mechanics model, mooring dynamics model, and LEG (linear electric generator) electro-magnetic model were independently developed and then fully coupled in time domain to account for the cross-coupling interactions among them. Then, the developed simulation tool was verified component by component against various laboratory tests. Subsequently, systematic parametric studies were conducted with several important design parameters under various wave conditions to enhance power generation. After that, the average output power was evaluated in enlarged operational wave conditions. The present SR-WEC is particularly designed to be efficient at low sea states, which is good since they cover the majority of typical annual sea states.

Keywords: Wave energy converter, surface riding, linear electric generator, mooring line, floater-mooring-generator interaction, time-domain simulation

1. Introduction

Renewable energy resources, such as wind, ocean wave, and solar energy, have been considered as an alternative and environmentally friendly option to replace conventional energy resources. They are expected to account for 47.7% of the total energy consumption in 2040 based on a global renewable energy scenario [1]. Among them, ocean waves are a promising energy resource because of its highest energy density (2-3 kW/m²), continuous availability (up to 90% of the time), and minimal negative environmental impact [2]. The global gross resource of the ocean wave is estimated to be in a range of 1-10 TW [3, 4].

In this regard, various WECs (wave energy converters) have been proposed. Among many technical aspects, the proper selection of the PTO (power take-off) system is important in improving conversion efficiency. Hydraulic systems (e.g. Pelamis [5]), turbines (e.g. Wave Dragon [6]), and direct-drive WECs (e.g. PowerBuoy [7]) are popular PTO systems. Various types

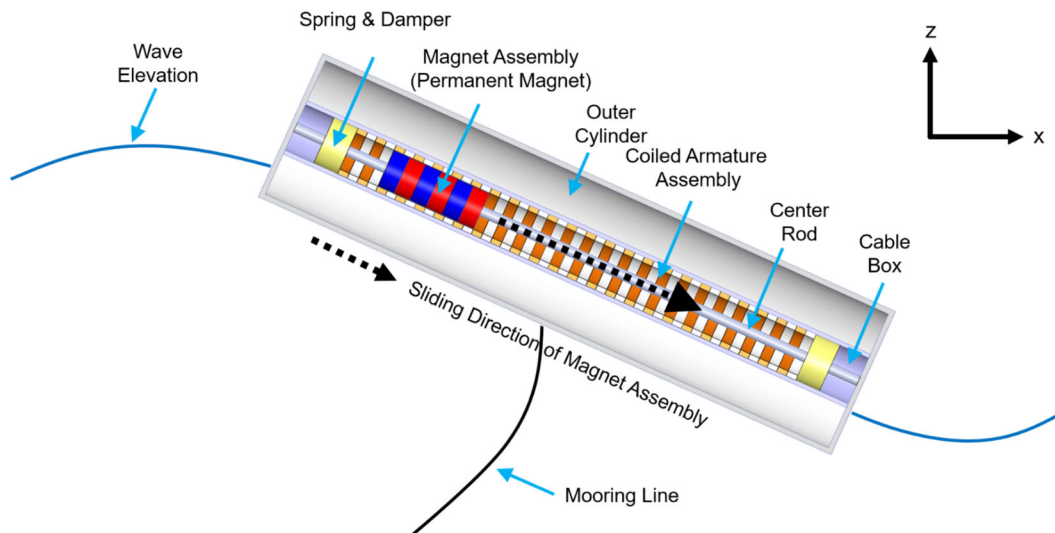
41 of WEC devices and their simulation methodologies were presented in Ref. [8]. The hydraulic
42 PTO system requires additional energy conversion to operate the rotary generator, which makes a
43 device complex. Additional energy losses are inevitable in the process of double energy
44 conversion. Also, the complexity of the conversion process can lead to reliability and maintenance
45 issues [9], especially in harsh ocean environments. On the other hand, direct-drive WECs with the
46 LEG (linear electric generator) produce power from the relative translational motions between the
47 permanent magnet and coiled armature. They do not require any intermediate step; thus, the design
48 can be simpler [2].

49 In recent decades, a number of experiments and numerical studies have been performed on
50 the direct-drive WECs. For example, Prudell et al. [10] devised a freely floating dual-buoy WEC
51 and conducted a laboratory experiment to estimate its performance. Kim et al. [11] designed a
52 dual-buoy WEC, which utilized three resonant motions of two bodies and moon-pool, and carried
53 out a laboratory experiment under regular and random wave conditions. Their experimental results
54 were also compared with the results of time-domain dynamics simulations under the random-wave
55 excitation [12]. Lejerskog et al. [13] conducted large-scale sea tests of the point absorber that was
56 mounted at the seabed of the Lysekil research site in Sweden. They concluded that higher output
57 power can be achieved in upward motions than downward motions. Stelzer and Joshi [14] also
58 studied a similar point absorber under random wave excitations using numerical simulations.
59 Zheng et al. [15] designed the variable aperture point absorber for enhancing survivability,
60 reducing cost, and improving wave power absorption. Linear and nonlinear hydrodynamic models
61 were compared for the two-body WEC [16] and submerged point absorber [17]. In addition,
62 various design-optimization strategies have been adopted to improve the performance of the direct-
63 drive WEC. Parametric studies were performed to find the proper generator and structural design
64 [18, 19]. Resonance of dual-body WECs was adjusted to maximize the relative motion by using
65 linear springs connected between two bodies [20-22]. The tuned inertia mass was utilized in a
66 point absorber to increase power absorption and broaden the effective wave frequency range [23].
67 Control systems, such as latching control [24] and model predictive control [25], were adopted to
68 improve the overall output power.

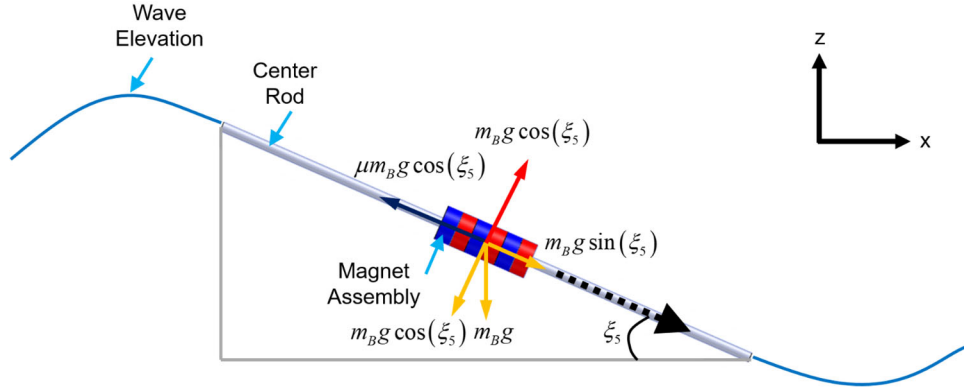
69 In this study, we developed a hydro-dynamics/linear-generator fully coupled simulation
70 program to investigate the performance of an innovative SR-WEC (Surface Riding WEC) devised
71 by authors [26]. It mainly consists of an outer cylinder equipped with a coiled armature assembly
72 and a magnet assembly sliding along a center rod (see Fig. 1). Compared to other direct-drive
73 WECs that utilize relative heave motions, SR-WEC uses the sliding motion inside a surface-riding
74 and pitching horizontal cylinder. It is hard to find similar concepts and relevant simulations in
75 publicly available literature [27]. The SR-WEC is specially devised to generate appreciable
76 electrical power even in low sea states, which typically cover 90 % of the annual sea state e.g. East
77 Coast in the U.S. (31.887 N 74.921 W). It is possible since both wave heights and lengths are
78 reduced in low sea states for wave slopes to remain about the same. SR-WEC generates the sliding
79 motion by the gravity acceleration, while most existing LEG-based devices utilize inertial
80 acceleration. Various optimization methods can be employed to enhance output power further.

81 To evaluate the SR-WEC's performance, the present time-domain simulation program was
82 developed and used to solve the fully-coupled floater-mooring-generator interaction. Several
83 major considerations and algorithms were made to develop the numerical model. First, the outer
84 cylinder interacts with waves, and thus its hydrodynamic coefficients and wave-excitation force
85 were estimated based on potential theory in frequency domain [28] and they were utilized in the
86 subsequent time-domain motion-simulation program. Second, the sliding mechanism of the
87 magnet assembly with time-varying contact and friction forces should be well understood and
88 modeled. Third, a reasonable collision model was developed at both ends of the outer cylinder to
89 assess the impact-induced velocity, load, and fatigue by the magnet assembly. Fourth, the PTO
90 force between the coil on the outer cylinder and the sliding magnet assembly should be well
91 estimated to correctly estimate the resisting force and generated power. Fifth, a single point
92 mooring system is installed and connected to the outer cylinder for a station-keeping purpose, and
93 the floater-mooring interaction should be solved. The numerical modeling of sliding mechanism
94 and collision at both ends was validated through comparisons with the physical system after
95 mounting and pitching the SR-WEC on harmonic actuators with the PTO off. The numerical
96 generator dynamics was also verified through comparisons with the laboratory experiments of
97 Prudell et al. [10] with PTO on. Afterward, a series of parametric studies were performed with the
98 validated numerical model to better understand the system's sensitivity on the design parameters
99 so that the results can be used as the improved design for given annual sea states.

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(a) Design



(b) Working principle

Fig. 1. Design and operation principle of the SR-WEC.

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2. Design and Operation Principle

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The design and operation principle of SR-WEC are presented in Fig. 1. The device mainly consists of the outer cylinder equipped with the armature assembly and the magnet assembly. The magnet assembly is composed of neodymium (NdFeB) magnets, supporting lamination steel, and linear ball bearings in detail. The linear ball bearings in the magnet assembly are in contact with a fixed center rod, which allows single-degree-of-freedom (SDOF) motion in the sliding direction along the center rod. The armature assembly, composed of coil and lamination steel, is installed to the outer cylinder and surrounds the magnet assembly. The outer cylinder interacts with the surrounding ocean fluid. The spring-damper system, which is referred to as the end damper, is located at both ends of the outer cylinder to alleviate collision damage from the magnet assembly. Ring masses are also located at both ends to provide a large mass moment of inertia about the y-axis (M_{55}) so that pitch natural frequency (1.65 rad/s) can be close to peak frequency in low sea states. As shown in Fig. 2, a single point mooring line is connected to the bottom center of the outer cylinder for position keeping, and cylinder's longitudinal direction is parallel to the dominant wave direction. Table 1 summarizes important design parameters.

Electric power is generated by relative motions between the magnet and armature assemblies. As shown in Fig. 1(b), the relative motion by sliding is mainly induced by the pitch motion of the outer cylinder. As the inclination angle is larger than the minimum sliding angle, the magnet assembly starts to slide by gravity. The detailed formulations are given in Section 3.

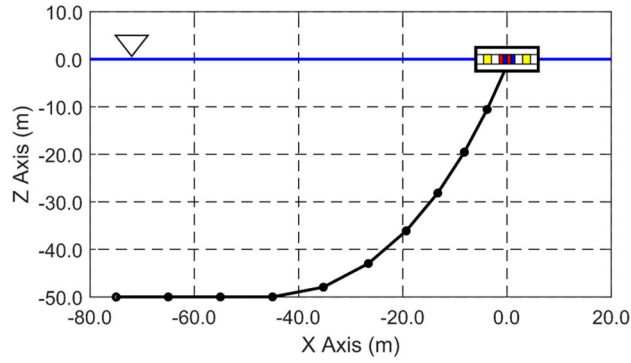


Fig. 2. Side view of SR-WEC with a mooring line.

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Table 1. Important parameters of SR-WEC.

Component	Item	Value	Unit
Outer cylinder	Length	8	m
	Diameter	2.6	m
	Mass	21368	kg
Magnet assembly	Length	1	m
	Diameter	0.38	m
	Mass	400	kg
	Air gap	0.5	cm
Linear electric generator	Phase resistance	4.58	Ω
	Phase inductance	190	mH
	Magnet pole pitch	72	mm
	Coil pitch	72	mm
Mooring line (Studlink Chain)	Nominal diameter	1.5	cm
	Length	100	m
	Mass/unit length	4.9	kg/m
	Minimum breaking load	263.9	kN
Mass matrix of SR-WEC	M_{11}, M_{22}, M_{33}	21768	kg
	M_{44}	18343	kg·m ²
	M_{55}, M_{66}	273508	kg·m ²

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3. Coupled Time-Domain Simulation

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A hydrodynamics-mechanics-generator fully-coupled time-domain simulation program was developed to assess the performance of the SR-WEC i.e. floater-mooring-generator interactions were solved at each time step to evaluate their dynamic responses, mooring tension, and electric output power. The time-domain dynamic analysis allows non-linear loads to be considered. Section 3 explains the theory and formulations of the time-domain simulation and coupling methods among the floater (i.e. outer cylinder), magnet assembly, and mooring line.

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3.1. Floating-Body Model

In time domain, the 6 DOF dynamic responses of the outer cylinder can be evaluated by solving the Cummins equation [29] as:

$$\begin{aligned} & (M_{ij} + A_{ij}^\infty) \ddot{\xi}_j(t) + B_{ij}^E \dot{\xi}_j(t) + K_{ij} \xi_j(t) \\ & = F_i^W(t) + F_i^C(t) + F_i^M(t) + F_i^G(t) + F_i^S(t) \quad i, j = 1, 2, \dots, 6 \end{aligned} \quad (1)$$

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where M_{ij} is the mass matrix, A_{ij}^∞ is the added mass matrix at the infinite frequency, B_{ij}^E is the external damping matrix, K_{ij} is the system's stiffness matrix induced by hydrostatic and gravitational stiffness, $\xi_j = [\xi_1, \xi_2, \xi_3, \xi_4, \xi_5, \xi_6]^T$ is the displacement vector, i.e., surge, sway, heave, roll, pitch, and yaw, F_i^W , F_i^C , and F_i^M are, respectively, the first-order wave-excitation, convolution, and Morison drag force vectors, F_i^G is the PTO force vector induced by the interaction between the magnet and armature assemblies, and F_i^S is the spring force vector to couple the outer cylinder with the mooring line. The upper dot in the equations represents the time derivative of a variable.

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Based on the assumption of linearity, A_{ij}^∞ , F_i^C , and F_i^W can be obtained from the equivalent relationship between the impulse-response-function-based equations in time domain and the diffraction/radiation-based equations in frequency domain as follows:

$$A_{ij}^\infty = A_{ij}(\omega) + \int_0^\infty R_{ij}(t) \frac{\sin(\omega t)}{\omega} dt \quad (2)$$

$$F_i^C(t) = -\int_0^\infty R_{ij}(\tau) \dot{\xi}_j(t-\tau) d\tau \quad (3)$$

$$R_{ij}(t) = \frac{2}{\pi} \int_0^\infty B_{ij}(\omega) \cos(\omega t) d\omega \quad (4)$$

$$F_i^W(t) = \text{Re} \left(\sum_{j=1}^N A_j^1 L_i(\omega_j) e^{i(k_j x - \omega_j t - \alpha_j)} \right) \quad (5)$$

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where R_{ij} is the retardation function, A_{ij} and B_{ij} are the added mass and the radiation damping as a function of angular frequency ω , and A_j^1 , L_i , k_j , and α_j are wave amplitude, the linear transfer function, wavenumber, and random-phase angle, respectively. Frequency-dependent A_{ij} , B_{ij} , and L_i were obtained by a 3D diffraction/radiation program [28].

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The Morison equation is widely used for the estimation of wave forces on slender bodies at its instantaneous position, which is composed of the linear inertia and non-linear drag terms [30]. The Morison equation is recommended for the wave-force evaluation of slender structures

165 when the ratio of outer diameter D to wavelength λ_w is less than 0.2 ($D / \lambda_w = 0.12$ at natural
 166 period=3.81 sec) [31]. Since the inertia term was obtained from the diffraction/radiation potential
 167 theory, only the non-linear drag term was added for the viscous-drag-force evaluation. The
 168 Morison drag force per unit length for the moving cylindrical body can be written as:

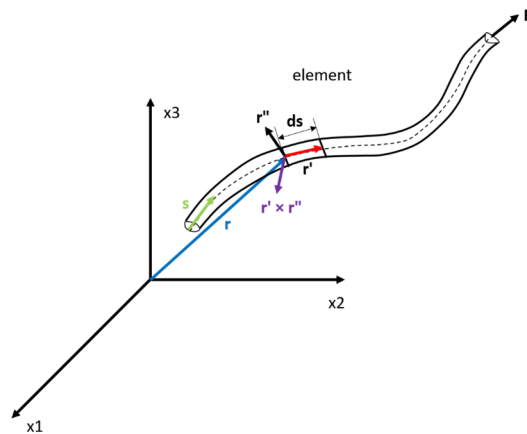
$$169 \quad F_i^M(t) = \frac{1}{2} \rho C_D D (u_i^n - \dot{\xi}_i^n) |u_i^n - \dot{\xi}_i^n| \quad (6)$$

171 where ρ is the density of water, C_D is the drag coefficient, and u_i is the surrounding fluid's
 172 velocity. Superscript n is the normal direction of a variable.

173 For SR-WEC, the mass of the magnet assembly was designed to be much lighter (1.9%)
 174 than that of the outer cylinder. In this regard, we evaluated the dynamic responses of the outer
 175 cylinder based on the total mass of the SR-WEC including the mass of the magnet assembly,
 176 instead of solving the two-body fully-coupled interaction between the outer cylinder and the
 177 magnet assembly.
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179 3.2. Mooring-Line Model and Coupling with Floating Body

180 As mentioned before, a single-point mooring line was employed to maintain SR-WEC's
 181 original position. Therefore, the outer cylinder's motions are to be coupled with the mooring
 182 dynamics. Two governing equations were established to estimate the mooring dynamics and
 183 tension based on a rod theory [32], in which generalized coordinate was used and twisting motions
 184 and moments were neglected. Fig. 3 presents the rod theory's coordinate system. The generalized
 185 coordinate system is along the line, and thus the geometric nonlinearity can be automatically
 186 considered [33]. In the generalized coordinate system, $r_i(s, t)$ represents the position vector in 3D,
 187 which is a function of arc length s and time t to define space curve, r_i' denotes the unit tangent
 188 vector to the space curve, while r_i'' and $r_i' \times r_i''$ are the principal normal and bi-normal vectors,
 189 respectively. The prime in the equations denotes the spatial derivative of a variable.
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192 Fig. 3. Rod theory's coordinate system [34].

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The equation of motion with the line's tension and bending effects can be described as follows:

$$197 \quad -\left(EI r_i''\right)'' + \left(\tilde{\lambda} r_i'\right)' + q_i^D = m_D \ddot{r}_i \quad i = 1, 2, 3 \quad (7)$$

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199 where E is Young's modulus, I is the second moment of cross-sectional area, $\tilde{\lambda} = \tilde{T} - EI\kappa^2$ with
200 effective tension \tilde{T} and local curvature κ , while q_i^D and m_D are the distributed force vector and
201 mass per unit length. In this case, q_i^D is the sum of wet weight vector per unit length \tilde{w}_i^R (
202 = $w_i^R + B_i^R$ where w_i^R and B_i^R are weight and buoyancy per unit length) and the wave force vector
203 per unit length F_i^D . According to the Morison equation, F_i^D for the cylindrical body can be written
204 as:

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$$206 \quad F_i^D = -C_A \rho A_E \ddot{r}_i^n + C_M \rho A_E \dot{u}_i^n + \frac{1}{2} C_D \rho D (u_i^n - \dot{r}_i^n) |u_i^n - \dot{r}_i^n| \quad (8)$$

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208 where C_A and C_M are, respectively, the added mass and inertia coefficients, A_E is the external
209 cross-sectional area. With \tilde{w}_i^R and F_i^D , Eq. (7) can be rewritten as:

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$$211 \quad m_D \ddot{r}_i + C_A \rho A_E \ddot{r}_i^n + (EI r_i''\right)'' - \left(\tilde{\lambda} r_i'\right)' = \tilde{w}_i^R + C_M \rho A_E \dot{u}_i^n + \frac{1}{2} C_D \rho D (u_i^n - \dot{r}_i^n) |u_i^n - \dot{r}_i^n| \quad (9)$$

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213 The mooring tension can separately be estimated by the extensible condition. Assuming
214 linear and small extension, the mooring tension can be estimated by the following relationship:

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$$216 \quad r_i' \cdot r_i' = \left(1 + \frac{T}{EA_T}\right) \approx 1 + 2 \frac{T}{EA_T} \approx 1 + 2 \frac{\lambda}{EA_T} \quad (10)$$

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218 where $A_T = A_E - A_I$ with the internal cross-sectional area A_I , T is the tension, and
219 $\lambda (= T - EI\kappa^2)$ is the Lagrange multiplier. The equation of motion and extensible condition in Eqs.
220 (9) and (10) are, therefore, governing equations for mooring-line analysis. Finally, Finite Element
221 (FE) formulations of the governing equations were further derived by using the Galerkin method,
222 which is detailed by Refs. [34, 35]. In the FE method, nonlinear behaviors can be captured by
223 dividing a line into multiple high-order elements.

224

225 A mooring line was coupled with the outer cylinder through translational and rotational
226 springs, which is a practical approach to connect several objects conveniently. The interaction
force is delivered to the outer cylinder as spring force, and equal and opposite force is also

227 transmitted to the top of the mooring line. Assuming the hinged connection, zero rotational
 228 stiffness was implemented. The spring force vector transmitted from the mooring line to the
 229 floating body can be expressed as [36]:

$$230$$

$$231 \quad F_i^S = \tilde{K}_{ij} (\tilde{T}_{jk} \tilde{u}_k^R - \tilde{u}_j^C) \quad (11)$$

232 where \tilde{K}_{ij} represents the coupling stiffness matrix, \tilde{T}_{jk} denotes the transformation matrix between
 233 the floater's origin and the connection position, while \tilde{u}_k^R and \tilde{u}_j^C are, respectively, the
 234 displacement vectors of the rigid body and connection position.
 235

236 3.3. Linear Generator Model

237 LEG was modeled to couple the two bodies and generate electric power. In the proposed
 238 design, the coiled armature assembly is attached to the outer cylinder whereas the magnet assembly
 239 freely slides along the center rod. The relative velocity between the armature and magnet
 240 assemblies induces the EMF (electromotive force), whose unit is voltage. According to Faraday's
 241 law of induction, the induced EMF E_b proportionally increases with a change in flux linkage
 242 $\lambda_{fl}(= N_c \phi_m)$ and can be calculated as [37]:
 243

$$244$$

$$245 \quad E_b = \frac{d\lambda_{fl}}{dt} = \frac{d\xi_7}{dt} \frac{d\lambda_{fl}}{d\xi_7} = \dot{\xi}_7 \frac{d\lambda_{fl}}{d\xi_7} \quad (12)$$

246 where N_c is the number of turns, ϕ_m is magnetic flux, and ξ_7 is the displacement of the magnet
 247 assembly relative to the armature assembly. The EMF is proportional to the relative velocity.
 248 Based on the RL (Resistor-Inductor) circuit, the following relationship can be used to compute the
 249 induced current [37]:
 250

$$251$$

$$252 \quad E_b = (R_L + R_C) \cdot i_c + L_P \frac{di_c}{dt} \quad (13)$$

253 where R_L and R_C are load and phase resistances, respectively, i_c is the induced electric current,
 254 and L_P is the phase inductance. The first-order ODE (ordinary differential equation) should be
 255 solved to acquire the induced current, and the fourth-order Runge-Kutta method was used.
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257 After calculating the induced current, the PTO force can be estimated by using the Lorentz-
 258 force equation. The electrons of electric current experience the magnetic force under the given
 259 magnetic field, which can be regarded as an interaction force between the two bodies. The PTO
 260 force on the coiled armature assembly in the sliding direction can be written as:
 261

$$262 \quad F^{Gn} = i_c \oint d\mathbf{l} \times \mathbf{B}_m = -B_m l_c i_c \quad (14)$$

263 where $B_m (= B_f \cos(\pi\xi_7 / \tau))$ with the magnitude of magnetic flux density B_f and pole pitch τ is
 264 the magnetic flux density, and l_c is the length of the coil that receives influences by the magnetic
 265 field at each time. According to Newton's third law, the equal and opposite force also acts on the
 266 magnet assembly. Since it acts in the sliding direction, at each time step, the PTO force was
 267 decomposed to calculate F_i^G for the outer cylinder. In general, B_f can be evaluated from electro-
 268 magnetic-filed simulation; however, in this study, a parametric study was conducted to find the
 269 proper B_f . In the parametric study, $B_f l_c$ is defined as the magnitude of EMF (or PTO force).
 270 Finally, the generated electric output power can be expressed as:

$$272 \quad P_o = i_c^2 R_L = i_c V_o \quad (15)$$

273 where V_o is the output voltage.

274 275 276 **3.4. Magnet-Assembly Dynamics**

277 Relative-displacement-based SDOF equation of motion for magnet assembly in sliding
 278 direction along the center rod can be written as:

$$279 \quad m_B \ddot{\xi}_7(t) = F^L(t) - F^{Gn}(t) \quad (16)$$

280 281 282 where m_B is the mass of the magnet assembly and F^L is the sliding force.

283 The working principle of the magnet assembly was previously described (Fig. 1b). As the
 284 inclination angle of the outer cylinder ξ_5 (i.e. pitch motion) is higher than the minimum sliding
 285 angle associated with the given friction coefficient and lubrication condition, the magnet assembly
 286 starts to slide by gravity along the center rod.

287 The SDOF was defined in the body-fixed coordinate system located to the center of gravity
 288 of the outer cylinder. Since the coordinate center of the magnet assembly continuously moves at
 289 each time step due to the dynamic motions of the outer cylinder, the non-inertia reference frame,
 290 i.e., the accelerated coordinate system, was introduced to always keep the magnet assembly's
 291 coordinate center with respect to the center of gravity of the outer cylinder. The SDOF motion is,
 292 therefore, that of the magnet assembly relative to the body-fixed coordinate system of the outer
 293 cylinder in the sliding direction. In this coordinate system, the inertial forces, which are also known
 294 as the fictitious forces, should be added as external force terms. The inertial force is the production
 295 of the mass of the magnet assembly and acceleration of the outer cylinder. Considering the head
 296 wave condition, sway motion is generally small; thus, inertia force from sway motion was
 297 neglected. Therefore, F^L , which includes inertial forces, can be formulated as:

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$$F^L(t) = m_B \left(g \sin \xi_5 + \ddot{\xi}_3 \sin \xi_5 - \ddot{\xi}_1 \cos \xi_5 \right) - \text{sgn}(\dot{\xi}_7) \mu m_B \left(g \cos \xi_5 + \ddot{\xi}_3 \cos \xi_5 + \ddot{\xi}_1 \sin \xi_5 \right) \quad (17)$$

where g is gravity acceleration and μ is friction coefficient.

The equation of motion given in Eq. (17) does not account for the contact mechanism between two bodies at both ends of the outer cylinder. To realize the contact mechanism at both ends of the outer cylinder, conservation of momentum with the partial elastic condition was used. Then, the displacement and velocity of the magnet assembly in the sliding direction after a collision can be evaluated. The conservation of momentum and the coefficient of restitution can be expressed as:

$$m_a v_a(t_n) + m_b v_b(t_n) = m_a v_a(t_{n+1}) + m_b v_b(t_{n+1}) \quad (18)$$

$$R_E = \frac{v_b(t_{n+1}) - v_a(t_{n+1})}{v_a(t_n) - v_b(t_n)} \quad (19)$$

where v_a and v_b are the velocities of the objects #1 and #2, m_a and m_b are the masses of object #1 and #2, and R_E is the coefficient of restitution. By combining Eqs. (18) and (19), the velocity of object #1 after the collision can be derived as

$$v_a(t_{n+1}) = \frac{m_a v_a(t_n) + m_b v_b(t_n) + m_b R_E [v_b(t_n) - v_a(t_n)]}{m_a + m_b} \quad (20)$$

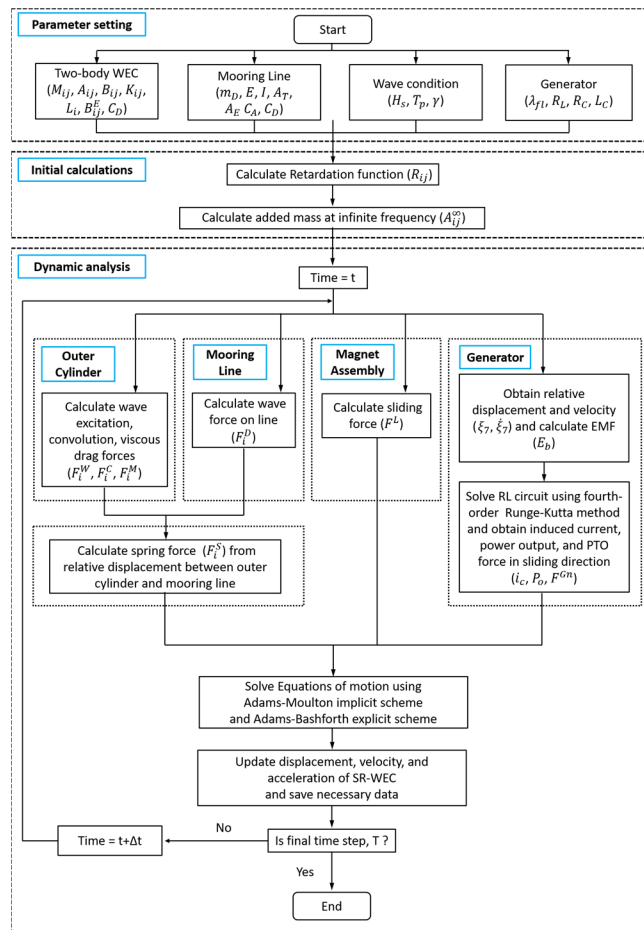
For SR-WEC, the mass of the outer cylinder is designed to be much larger than that of the magnet assembly; therefore, the terms related to the mass of the magnet assembly in Eq. (20) can be neglected. Moreover, for the same reason, the contribution of a collision to the outer cylinder's motion is assumed to be small, and the velocity of the outer cylinder is not influenced by the collision. Also, the non-inertia reference frame keeps the velocity of the outer cylinder zero from the view of the magnet assembly. Based on the above considerations, Eq. (20) can be simplified with the current variables of SR-WEC as:

$$\dot{\xi}_7(t_{n+1}) = -R_E \dot{\xi}_7(t_n) \quad (21)$$

When the magnet assembly does not contact both ends, Eq. (16) was used to solve the dynamic equation of motion. After the contact, Eq. (21) gave the instantaneous displacement and velocity of the magnet assembly.

3.5. Procedure of Time-Domain Simulation

334 Fig. 4 presents a flowchart describing the procedure of the present fully-coupled floater-
 335 mooring-generator time-domain simulation. First, the frequency-domain hydrodynamic
 336 computation was completed to obtain the frequency-dependent hydrodynamic coefficients and
 337 forces for the outer cylinder. Also, the initial parameters of the two bodies (outer cylinder and
 338 magnet assembly), mooring line, and generator were entered. Second, the retardation function and
 339 the added mass at infinite frequency were calculated. Third, at each time step, forces on two bodies
 340 and a mooring line were evaluated, which includes the wave, PTO, spring, and sliding forces.
 341 Fourth, the coupled equations of motion were solved by the Adams-Moulton implicit method
 342 combined with the Adams-Bashforth explicit method, in which iteration within a time step is not
 343 needed. The third and fourth steps were repeated until the last time step.
 344



345 Fig. 4. Flowchart describing the time-domain simulation.
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347 4. Validation of Numerical Model

348 Two comparative studies between the numerical simulation and experiment were carried
 349 out to validate the proposed simulation program: (i) the sliding mechanism of the magnet assembly
 350 without the LEG was confirmed by comparing the numerical simulations with the heave-pitch
 351 coupled actuator tests conducted by authors, and (ii) LEG dynamics by the present numerical

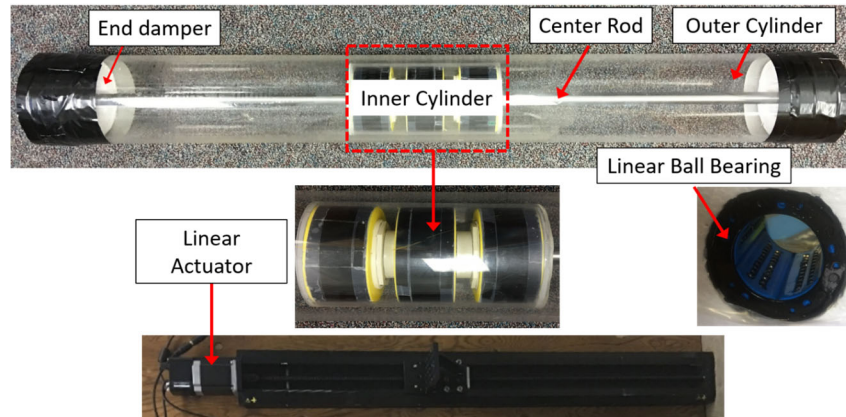
352 simulation at different sea states was validated against laboratory tests conducted by Prudell et al.
353 [10].

354

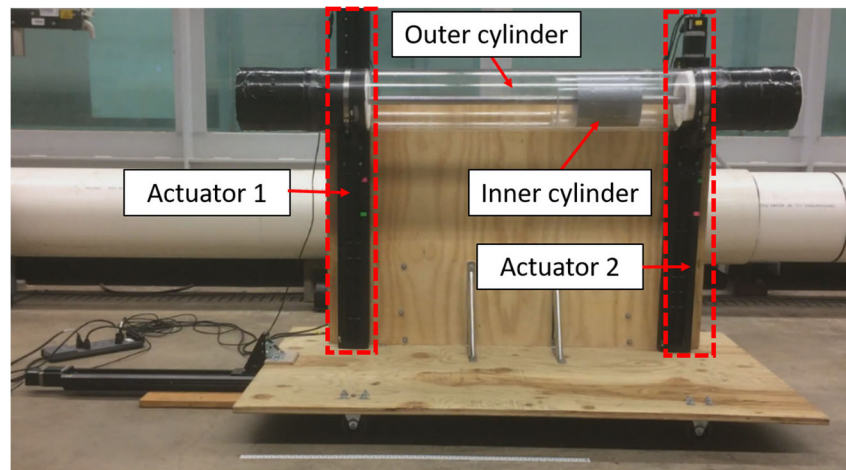
355 4.1. Validation of Sliding Mechanism

356 Fig. 5 shows the test setup of the heave-pitch coupled actuator test. Table 2 presents the
357 dimension and mass of cylinders. An acrylic hollow tube was used for inner and outer cylinders
358 while a center rod was made of stainless steel. A linear ball bearing (SSU20 by Thomson) was
359 mounted to an inside of the inner cylinder to maximize the sliding performance with a low friction
360 coefficient. At both ends of the outer cylinder, Styrofoam dampers were located to mitigate the
361 impact force. There were two actuators (A-LST series by Zaber Technology) vertically installed
362 at both ends of the outer cylinder so that the heave-pitch coupled motion of the outer cylinder can
363 be emulated. As the outer cylinder moves with the actuators, the inner cylinder slides along the
364 center rod. For this test, LEG was not installed since the main objective of the test was to check
365 the sliding mechanism.

366



(a) Components and equipment



(b) Test setup

367

Fig. 5. Test setup of heave-pitch coupled actuator tests.

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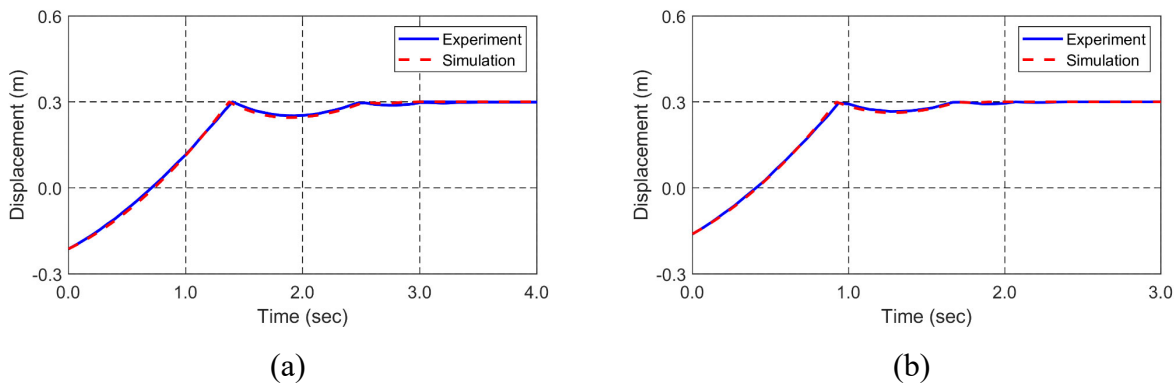
Table 2. Scaled WEC's mass and dimension.

Item		Value
Outer cylinder	Length	1.52 m
	Mass	9.51 kg
Inner cylinder	Length	0.3 m
	Mass	2.1 kg
Center rod	Outer diameter	0.025 m

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First, the minimum sliding angle of the inner cylinder was checked. The inclination angle was increased from 0.2 degrees with an interval of 0.05 degrees, which results in the observed minimum sliding angle of around 0.5 degrees. The corresponding static friction coefficient was estimated from the minimum sliding angle.

Second, the dynamic friction coefficient along the sliding and collision mechanism at the ends were identified at different inclination angles. Fig. 6 shows the displacements of the inner cylinder in the sliding direction at inclination angles of 2 and 3 degrees. The dynamic friction coefficient is generally smaller than the static friction coefficient, and the identified dynamic friction coefficient is 75 % of the static friction coefficient to match the experimental results before a collision occurs. Also, the coefficient of restitution was checked by comparing the transient response after the collision. As shown in Fig. 6, the coefficient of restitution in the range of 0.36 ~ 0.39 best matches against the experiments. Those values were used in the ensuing simulation of SR-WEC.

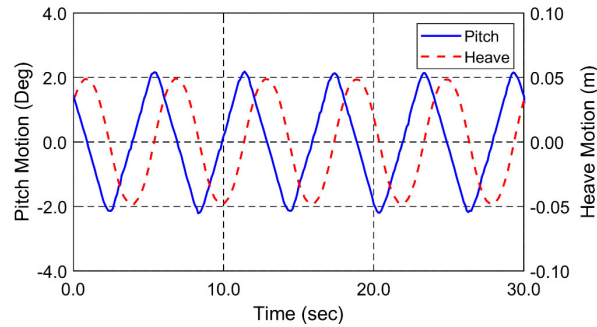


385 Fig. 6. Time-history comparison of the displacement of the inner cylinder in the sliding direction
386 at the inclination angles of 2 (a) and 3 (b) degrees.

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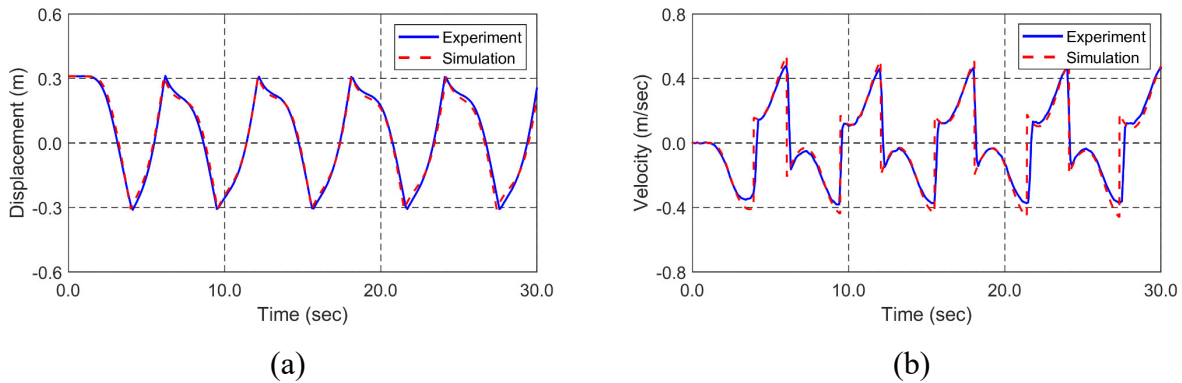
Third, a heave-pitch coupled actuator test was performed, and the experimental results were compared with our simulation results. Fig. 7 shows the measured time-histories of the outer cylinder's pitch and heave motions, and the corresponding displacements and velocities of the inner cylinder are presented in Fig. 8. The same condition was also inputted in the numerical

392 simulation. As shown in Fig. 8, with the coefficient of restitution of 0.38, the simulated magnitudes
 393 and trends of displacement and velocity coincide well against experimental values. Those well-
 394 matched sliding responses with the experiments, where the collision and those two coupled
 395 motions were involved, validate the formulation of the SR-WEC mechanics as explained in the
 396 previous section.
 397



398 Fig. 7. Time-history of measured pitch and heave motions.

399



400 Fig. 8. Time-history comparison of displacement (a) and velocity (b) of the inner cylinder in the
 401 sliding direction for the heave-pitch coupled actuator tests.

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4.2. Validation of Generator Dynamics

The developed LEG simulation model was validated with the experimental data given in Prudell et al. [10]. The present formulations related to generator dynamics were confirmed through this validation. In the experiment, two-body heave-type WEC with LEG was tested in the laboratory. Armature and magnet assemblies were installed at inner and outer buoys, respectively. 960 NdFeB magnets were used to construct magnet assembly, and the three-phase Y connection was designed. The major LEG parameters are summarized in Table 3. Considering that the LEG performance only depends on the relative displacement and velocity, the difference in WEC type is irrelevant as long as those two inputs are the same.

Table 3. Major parameters for calculation of electric output power [10].

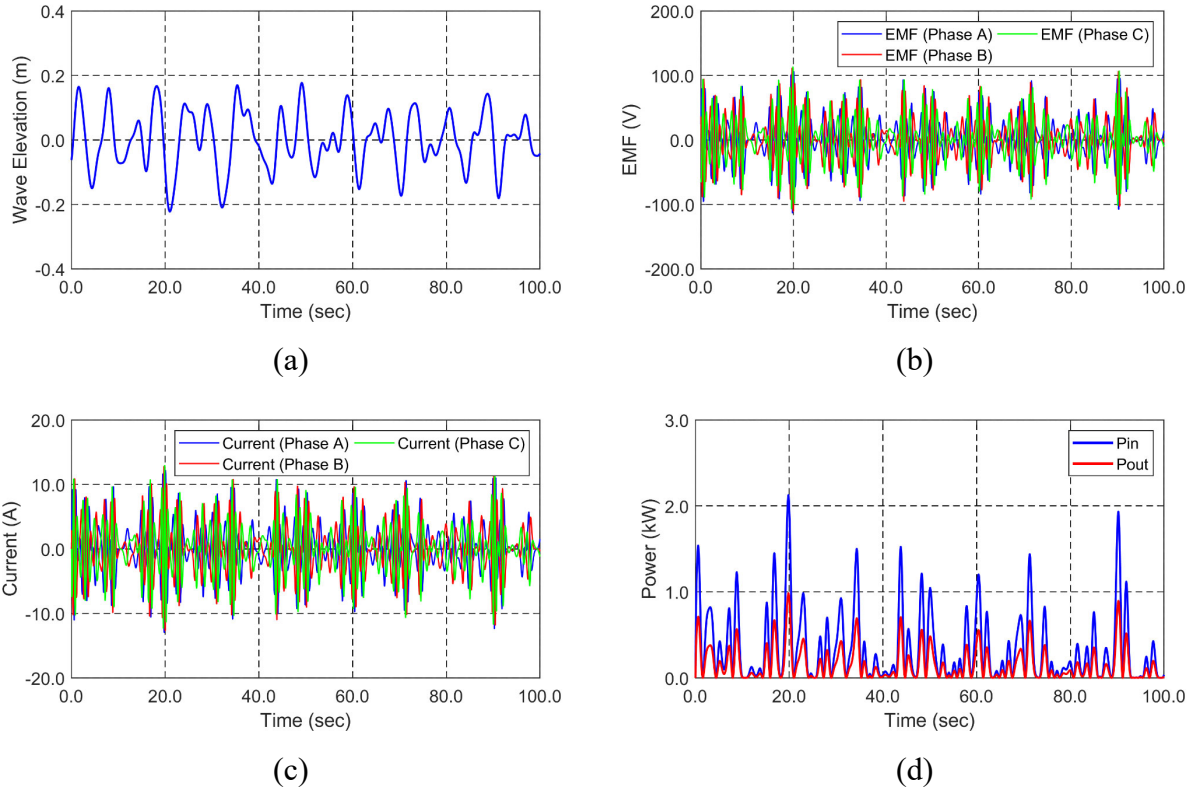
Parameter	Value	Units
Number of magnetic poles	4	-
Magnetic flux density	0.9037	T
Average circumference of the winding	1.81	m
Number of turns per slot	77	-
Reduction factor due to armature reaction	0.904	-
EMF magnitude at unit velocity	455.43	V
Phase resistance	4.58	Ω
Phase inductance	0.19	H
Wire diameter	1.628	mm

414

415 In their laboratory test, the dynamics of two floating bodies in waves was actually not
 416 considered. Instead, they assumed that the inner buoy is stationary while the outer buoy moves
 417 with wave elevation. Under this scenario, the heave-motion time series of the outer buoy is the
 418 same as wave-elevation time series while the inner buoy is fixed. Therefore, they actually utilized
 419 the measured time-histories of wave elevation as the relative heave motion between the two buoys.
 420 Then, they recorded the measured output power from the physical LEG for 900 sec for 8 different
 421 sea states. In the present numerical simulation, authors also generated the wave-elevation time
 422 histories for 900 sec for the same sea states and used them as the input relative displacement of the
 423 same numerical LEG. The wave-elevation time series were generated by superposing 100 regular
 424 waves from the PM (Pierson-Moskowitz) spectrum.

425 Fig. 9 shows the time-history examples of wave elevation, the EMF, the induced current,
 426 and input and output powers obtained by our numerical simulation in the case of significant wave
 427 height $H_s=0.44$ m, zero-crossing period $T_z=6.4$ sec, and load resistance $R_L=3.9 \Omega$. The time
 428 histories are presented for the first 100 seconds. The LEG-related frequencies are higher than those
 429 of wave elevation. In other words, the generator dynamics solver requires smaller time steps than
 430 the floating-body dynamics solver. The 120-degree phase difference of the three-phase system was
 431 also confirmed in the time histories. As summarized in Table 4, the calculated average output
 432 powers are well-matched with those from experiments i.e. their maximum difference is only 5.2
 433 %. The results in Table 4 demonstrate that the developed numerical LEG solver can be utilized for
 434 the ensuing SR-WEC simulations.

435



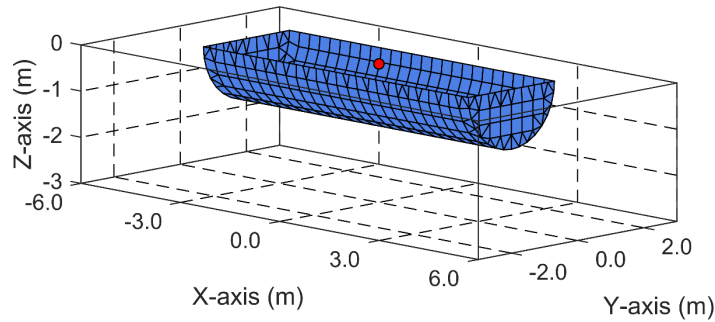
436 Fig. 9. Time histories of wave elevation (a), EMF (b), induced current (c), and input and output
 437 powers (d) for significant wave height of 0.44 m, zero-crossing period of 6.4 sec, and load
 438 resistance of 3.9Ω .

439 Table 4. Comparison of average output power between numerical and physical LEGs in various
 440 sea conditions [10].
 441

Sea condition	H_s (m)	T_Z (sec)	R_L (Ω)	Average output power (kW)		Percentage difference (%)
				Experiment	Simulation	
1	0.44	6.4	3.9	0.177	0.179	1.1
2	0.64	6.2	4.1	0.368	0.387	5.0
3	1.02	7.6	4.3	0.669	0.658	1.7
4	1.25	7.6	4.4	0.920	0.917	0.3
5	1.52	7.6	4.7	1.237	1.224	1.1
6	2.03	7.6	5.2	1.758	1.734	1.4
7	2.54	7.6	5.8	2.207	2.141	3.0
8	3.04	7.6	6.4	2.587	2.455	5.2

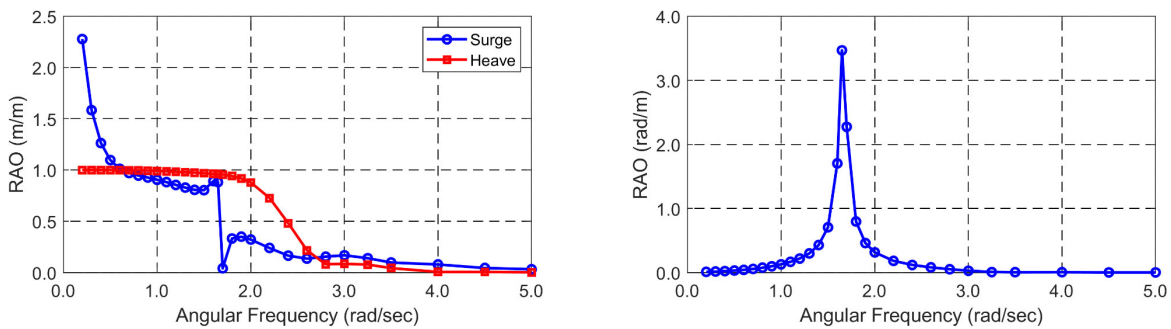
442
 443 **5. Results and Discussions**
 444 **5.1. Frequency-Domain Simulation**

445 The frequency-domain simulation was firstly conducted to compute the hydrodynamic
 446 coefficients and forces on the outer cylinder. A 3D diffraction/radiation program, WAMIT [28],
 447 was used to estimate the frequency-dependent added mass, the radiation damping coefficient, and
 448 the first-order wave-excitation force. These outputs were utilized in the time-domain simulation.
 449 SR-WEC is half-submerged, and the submerged surface panels were modeled, as shown in Fig.
 450 10. 30 wave frequencies were selected from 0.1 rad/s to 5.0 rad/s, and the wave direction was
 451 parallel to the longitudinal direction of SR-WEC. The mass matrix is based on the entire SR-WEC
 452 (See Table 1), including the magnet assembly located at the center of gravity of SR-WEC.
 453



454 Fig. 10. Panel model of the wet surface of the outer cylinder with 468 panels.

455 Fig. 11 shows the resulting surge, heave, and pitch Response Amplitude Operators (RAOs).
 456 After convergence test with different panel numbers as in the previous study [38], the converged
 457 results are given in Fig.11. The pitch natural frequency is 1.65 rad/s (3.81 sec), and thus excellent
 458 sliding performance is expected at low sea states. The computed added masses and radiation
 459 damping coefficients for the 3-DOF motions are plotted in Fig. 12.
 460
 461

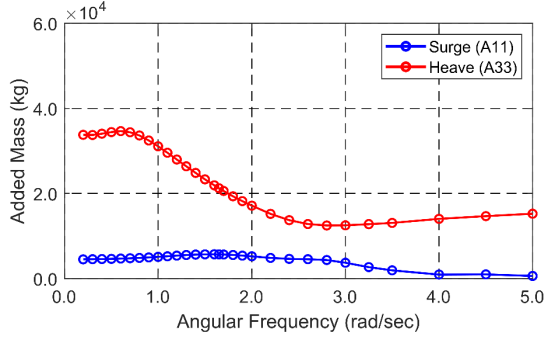


(a) Surge and heave RAOs

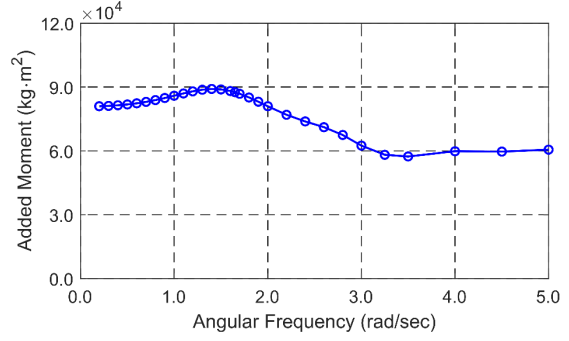
(b) Pitch RAO

462 Fig. 11. Surge, heave (a), and pitch (b) RAOs from 3D diffraction/radiation program.

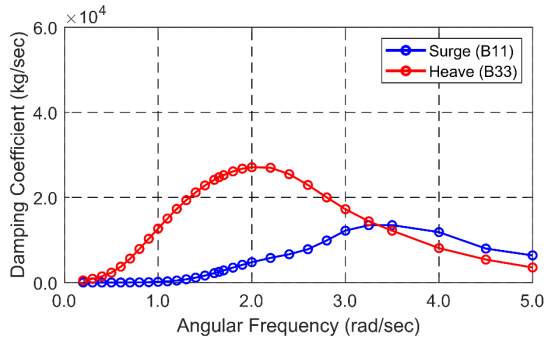
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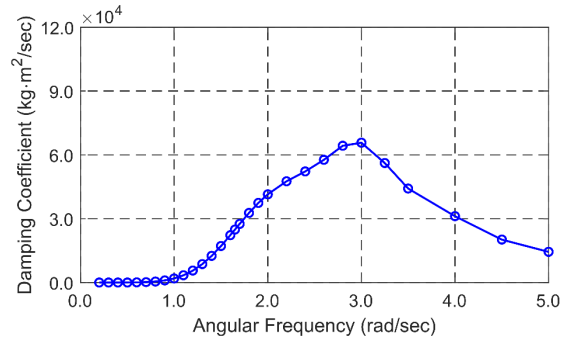
(a) A_{11}, A_{33}



(b) A_{55}



(c) B_{11}, B_{33}



(d) B_{55}

464 Fig. 12. Surge-heave-pitch added mass (a-b) and radiation damping (c-d) coefficients.

465

466 5.2. Time-Domain Simulation

467 The floater-mooring-LEG coupled dynamics simulations were performed in time domain
 468 for the performance estimation of SR-WEC. First, the LEG model was established through the
 469 initial parametric study with variable magnitudes of EMF and load resistance. A single-phase
 470 winding was employed in this study. Other generator parameters were the same as the parameters
 471 given in Prudell et al. [10]. Second, the magnet assembly's travel length, the coefficient of
 472 restitution at both ends, and the mass of the magnet assembly were selected for enhancing SR-
 473 WEC's performance. After each stage, the selected parameters were used as fixed parameters for
 474 the next parametric study. Last, the power-generation performance was quantitatively evaluated
 475 with the selected parameters under various random-wave excitations.

476 The time-domain model accounts for the viscous drag force on the outer cylinder. For the
 477 translational motions, the Morison equation was utilized with a drag coefficient of 0.5 acting at
 478 the outer cylinder's center of gravity, taking advantage of reference value in the experimental study
 479 at the representative surface roughness and Reynolds number [39]. For the rotational motions, a
 480 viscous damping ratio of 3% was assumed and inputted to the external damping matrix B_{ij}^E . The
 481 time step in the time-domain simulation was 0.005 sec to accommodate the generator dynamics
 482 accurately.

483

484 **5.2.1. Environmental Condition**

485 The JONSWAP wave spectrum was used for generating the time history of random waves.
 486 The range of significant wave heights was from 1 to 3.5 m while that of peak periods was from 4
 487 to 11 sec. The time history of wave elevations was generated by the superposition of 100 regular
 488 waves, and signal repetition was prevented by the adoption of randomly perturbed frequency
 489 intervals. The total simulation time for each case was 1200 seconds, for which the ramping time
 490 of 300 seconds was not included in the statistical assessment. The enhancement parameter γ in
 491 the JONSWAP wave spectrum was estimated by the proposed equation in Ref [40]:

492

$$\begin{aligned}
 \gamma &= 5, & \text{on } T_p / \sqrt{H_s} &\leq 3.6 \\
 \gamma &= \exp(5.75 - 1.15T_p / \sqrt{H_s}), & \text{on } 3.6 < T_p / \sqrt{H_s} < 5 \\
 \gamma &= 1, & \text{on } 5 \leq T_p / \sqrt{H_s}
 \end{aligned}
 \tag{22}$$

493

494 **5.2.2. Parametric Study 1: Magnitudes of EMF and Load Resistance**

495 Magnitudes of EMF (or magnitudes of PTO force i.e. term $B_f l_c$) and load resistance were
 496 first considered to design the PTO system. The PTO force acting on both the magnet and armature
 497 assemblies is a function of the induced current as given in Eq. (14), and the induced current is also
 498 related to the magnitudes of EMF, phase and load resistances, and phase inductance. Then, it is
 499 important to observe the relationship between these parameters and output power. In other words,
 500 the proper selection of these parameters can provide excellent sliding performance and output
 501 power simultaneously. Considering that EMF is a function of the relative velocity and the relative
 502 velocity also results from the PTO force, which is induced by the load resistance, the following
 503 parametric study can help in clarifying the role of LEG parameters to find improved output power.
 504

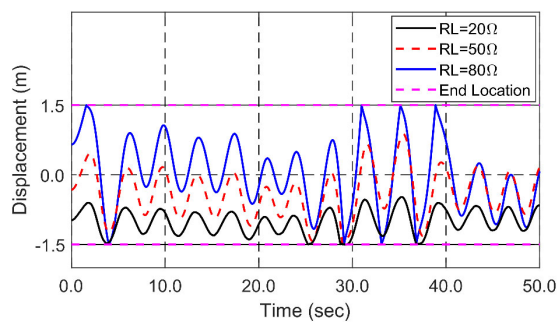
505 In this parametric study, load resistance and the magnitude of EMF were in the range of 10
 506 Ω to 200 Ω (10 Ω interval) and 100 T·m to 500 T·m (100 T·m interval), respectively. During the
 507 parametric studies, 3 wave conditions were selected, as summarized in Table 5. The sliding length,
 508 the coefficient of restitution, and the mass of the magnet assembly were fixed at 3 m, 0.38 (i.e.,
 509 obtained in the actuator test), and 400 kg, respectively.

510
 511 Table 5. Wave conditions for parametric studies.

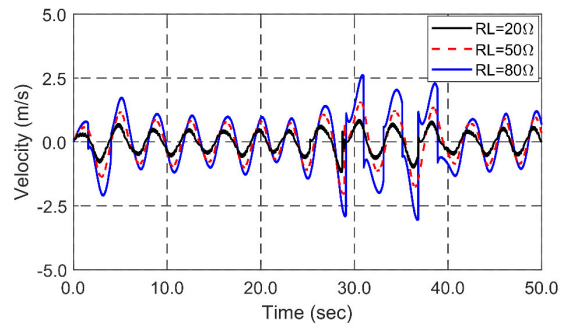
H_s (m)	T_p (sec)	γ	Mean wave slope (deg)
1.0	5.0	1.0	6.4
2.0	6.0	2.4	7.2
3.0	7.0	3.0	7.5

512
 513 Fig. 13 shows the time histories of displacement and velocity of the magnet assembly in

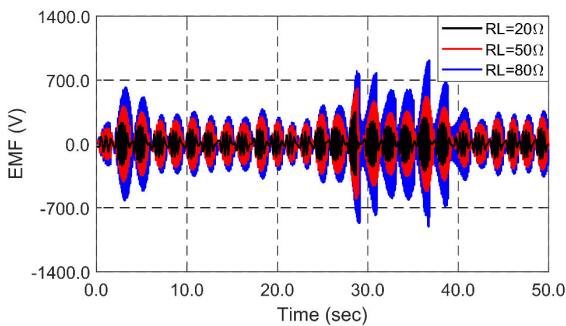
514 the sliding direction, EMF, induced current, output power, and pitch motion of the outer cylinder
 515 at various load resistances and for $H_s=2\text{m}$. In Fig. 13, we show the simulation results for three
 516 representative load resistances of 20, 50, and 80 Ω to explain results better with time histories
 517 while the magnitude of EMF was set to 300 T·m. The sliding performance varies with different
 518 load resistances, as shown in Fig. 13(a-b). Both sliding displacements and velocities increase with
 519 the increasing load resistance. The trend of EMF is similar to that of relative velocity i.e. larger
 520 EMF at larger load resistance, as shown in Fig. 13(c). However, the current is inversely
 521 proportional to load resistance, as represented in Fig. 13(d). The smallest induced current in the
 522 circuit contributes to the reduction of the PTO force. As a result, the highest output power is
 523 acquired at the load resistance of 80 Ω , as shown in Fig. 13(e). Time histories for other load-
 524 resistance cases over 80 Ω were also checked. The results show that even if the sliding performance
 525 is better, the higher output power is not observed owing to a large reduction in the current that is
 526 a source of output power. Moreover, it is confirmed that the motion of the magnet assembly is
 527 highly influenced by the pitch motion of the outer cylinder when Fig. 13(a-b) is compared with
 528 Fig. 13(f). In comparison between Fig. 13 (a) and (e), it turns out that the given design has
 529 significant power generation when the reaction velocity occurs from the collision.
 530



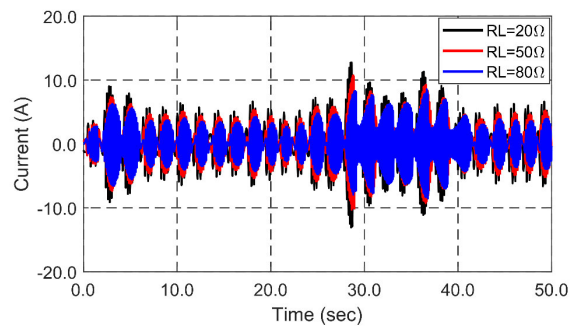
(a)



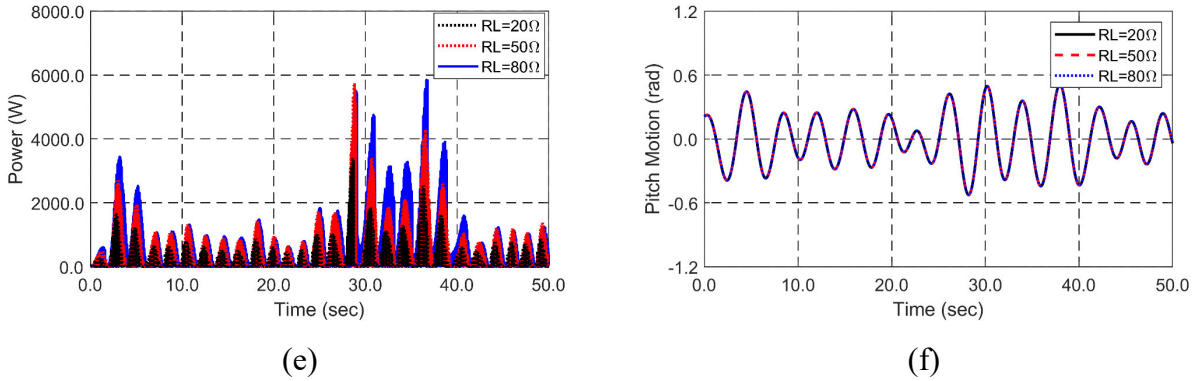
(b)



(c)



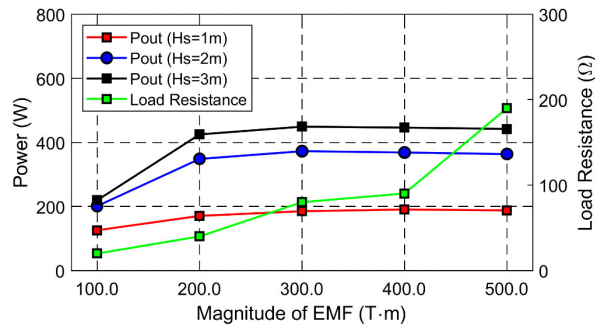
(d)



531 Fig. 13. Time histories of displacement (a) and velocity (b) of the magnet assembly, EMF (c),
 532 induced current (d), output power (e), and pitch motion of outer cylinder (f) at different load
 533 resistances ($H_s=2$ m, $T_p=6$ sec).

534
 535 Fig. 14 shows the average output power and load resistance at different magnitudes of EMF
 536 and wave conditions. The average output power is obtained from the time histories measured for
 537 900 s. The observed load resistance with the highest output power at each magnitude of EMF is
 538 the same regardless of wave conditions. The larger the magnitude of EMF, the higher the load
 539 resistance. Besides, when the magnitude of EMF is higher than 300 T·m, there is no change in
 540 output power as the magnitude of EMF increases, and this phenomenon is observed regardless of
 541 the sea state. The magnitude of EMF and load resistance that produce the maximum average output
 542 power are 300 T·m and 80 Ω, respectively. These selected LEG parameters were to be fixed for
 543 the next parametric study.

544



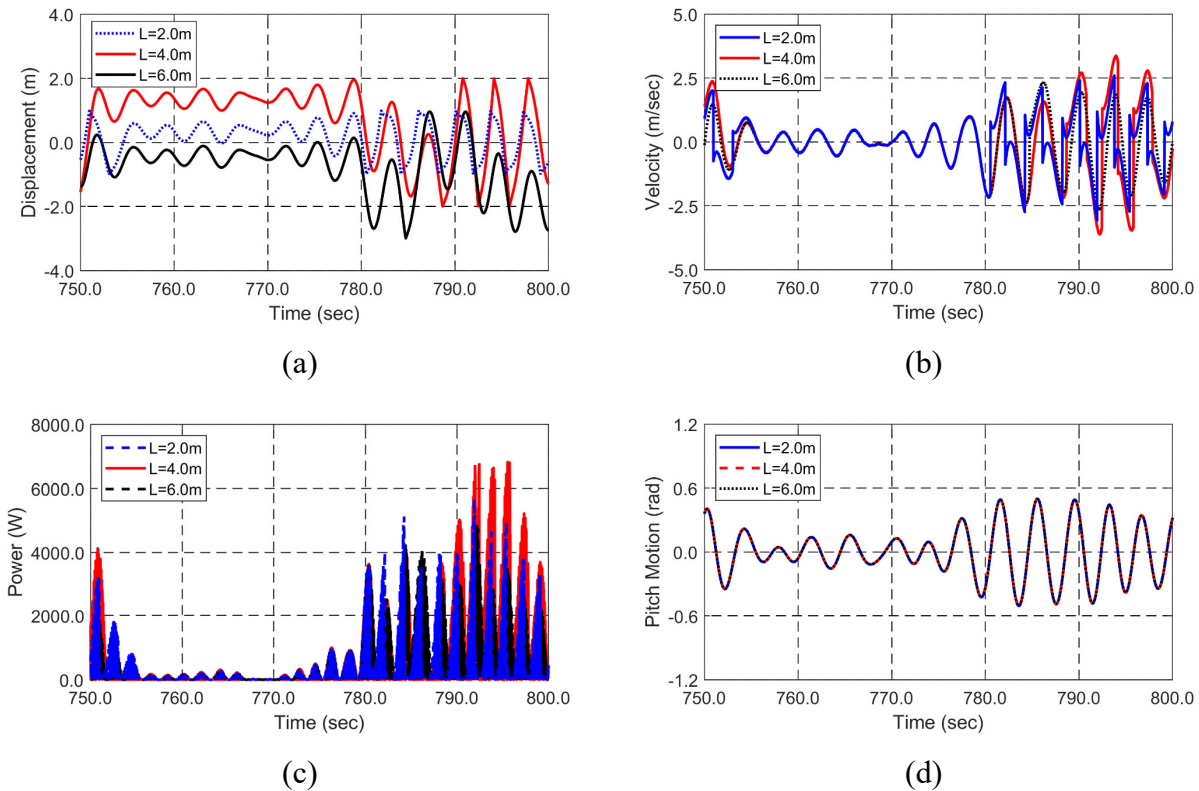
545 Fig. 14. Average output power and load resistance providing the maximum average output
 546 power at different magnitudes of EMF and wave conditions.

547

548 5.2.3. Parametric Study 2: Travel Length

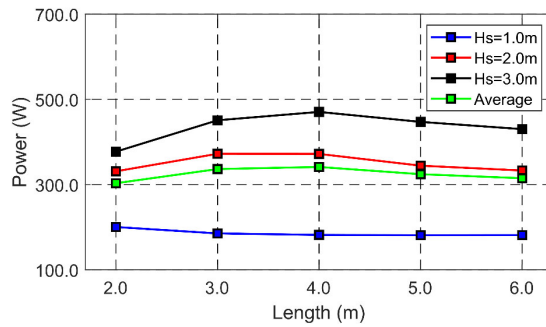
549 The travel length of the magnet assembly, defined as the length between two ends minus
 550 the length of the magnet assembly, can improve the sliding performance. Various lengths of 2 m
 551 to 6 m were selected with 1-m interval and the proper length was evaluated. The coil is installed

552 at the full length. Fig. 15 shows the time histories of displacement and velocity of the magnet
 553 assembly in the sliding direction, output power, and pitch motion of the outer cylinder at the
 554 different travel lengths. The magnet assembly can slide until it reaches both ends. In Fig. 15(a-b),
 555 interestingly, the magnet displacements and velocities become maximum when the travel length
 556 is 4 m. When the travel length is 2 m, the magnet assembly has to stop before it reaches the
 557 potentially maximum velocity. When the travel length is 6 m, the magnet does not slide the full
 558 length under the given wave condition as in Fig.15, so there is no improvement in output power.
 559 The magnet assembly gradually stops due to its inertial forces even after the pitch motion of the
 560 outer cylinder is switched in another direction. In this case, it can miss the best moment to slide in
 561 the other direction. Therefore, for the given wave conditions, the travel length at which average
 562 power becomes the highest is 4 m, as presented in Fig. 15(c). Moreover, the timely collisions
 563 at the ends are beneficial in increasing magnet sliding velocity, as in the case of travel lengths of 2
 564 m and 4 m. Furthermore, significant variations between cases are observed after contacts at both
 565 ends occur under the large pitch motion of the outer cylinder (780-800 sec). When there is no
 566 contact under the small pitch motion (755-780 sec), there is a minor difference in velocity since
 567 other parameters and the pitch motions are the same among them.
 568



569 Fig. 15. Time histories of displacement (a) and velocity (b) of the magnet assembly, output
 570 power (c), and pitch motion of outer cylinder (d) at different travel lengths ($H_S=2$ m, $T_P=6$ sec).

572 Fig. 16 presents average output powers at different travel lengths and wave conditions. In
 573 the case of $H_s=2$ m ($T_p=6$ sec) and $H_s=3$ m ($T_p=7$ sec), the output power does not increase even
 574 though the travel length is increased beyond 4 m; instead, the output power rather starts to decrease
 575 after travel length=4 m. Therefore, travel length=4 m can be considered as the appropriate travel
 576 length after averaging the output powers of the 3 sea states. The case is also benefited by timely
 577 end collision. Higher significant wave heights generate greater pitch motions, which increases the
 578 sliding performance and output power. However, when the pitch motion is much larger than that
 579 of current environmental conditions, the design with the travel length of 6 m may be able to
 580 generate higher output power, which is discussed in Section 5.2.6. If the travel length is too short,
 581 the magnet assembly quickly reaches one end and remains almost stationary until it slides in the
 582 other direction. So, the corresponding power-generation efficiency becomes relatively low. The
 583 travel length was fixed to be 4 m for the next parametric study.
 584



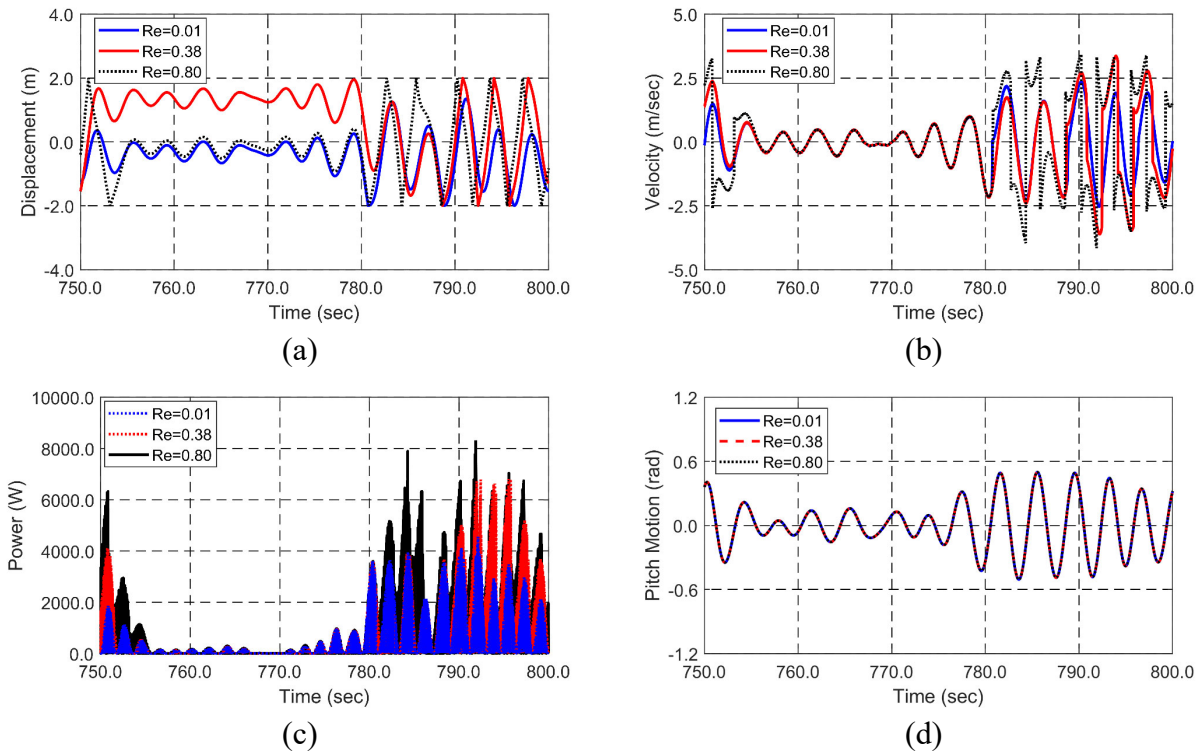
585 Fig. 16. Average output power at different travel lengths and wave conditions.

586
 587 **5.2.4. Parametric Study 3: Coefficient of Restitution**

588 It was found in the previous parametric study that the timely elastic collision at both ends
 589 (+2 m and -2 m) helps the magnet assembly to better slide with rebounding velocities. To confirm
 590 the improvement of the sliding performance due to the reactive velocity from the collision, we
 591 analyzed the effect of the coefficient of restitution on the average output power. In this regard,
 592 three restitution coefficients, 0.01, 0.38 (given by the actuator test), and 0.8 were considered.

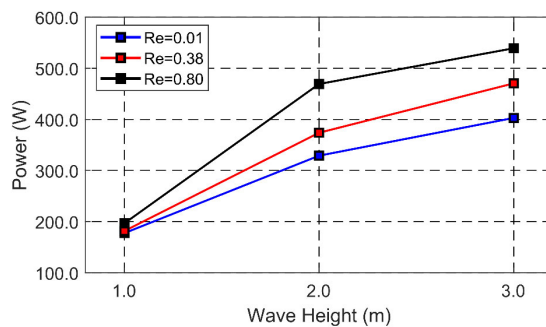
593 Fig. 17 shows the time histories of displacement and velocity of the magnet assembly in
 594 the sliding direction, output power, and pitch motion of the outer cylinder with the three different
 595 coefficients of restitution. Again, we observe the contact-induced substantial variations under the
 596 different coefficients of restitution when there is a significant pitch motion of the outer cylinder
 597 (780-800 sec). There is no contact when pitch motion is small (i.e. 750-780 sec), which leads to a
 598 minor difference in the sliding velocities since other parameters and the pitch motions are the same
 599 among them. When the timely collisions happen at the last stage of $Re=0.38$ and 0.8 cases (780-
 600 800 sec), the sliding velocities are increased due to the beneficial rebounding velocities after
 601 collisions. The higher the coefficient of restitution, the better the sliding performance and higher
 602 output power. This implies that the sliding performance can further be enhanced by placing highly
 603 restitutive elastic springs at both ends. Fig. 18 presents the average output power at different

604 coefficients of restitution and wave conditions, and the above trends can further be confirmed. For
 605 the following parametric study, the coefficient of restitution was to be fixed at 0.8.
 606



607 Fig. 17. Time histories of displacement (a) and velocity (b) of the magnet assembly, output
 608 power (c), and pitch motion of outer cylinder (d) at different coefficients of restitution ($H_S=2$ m,
 609 $T_P=6$ sec).

610



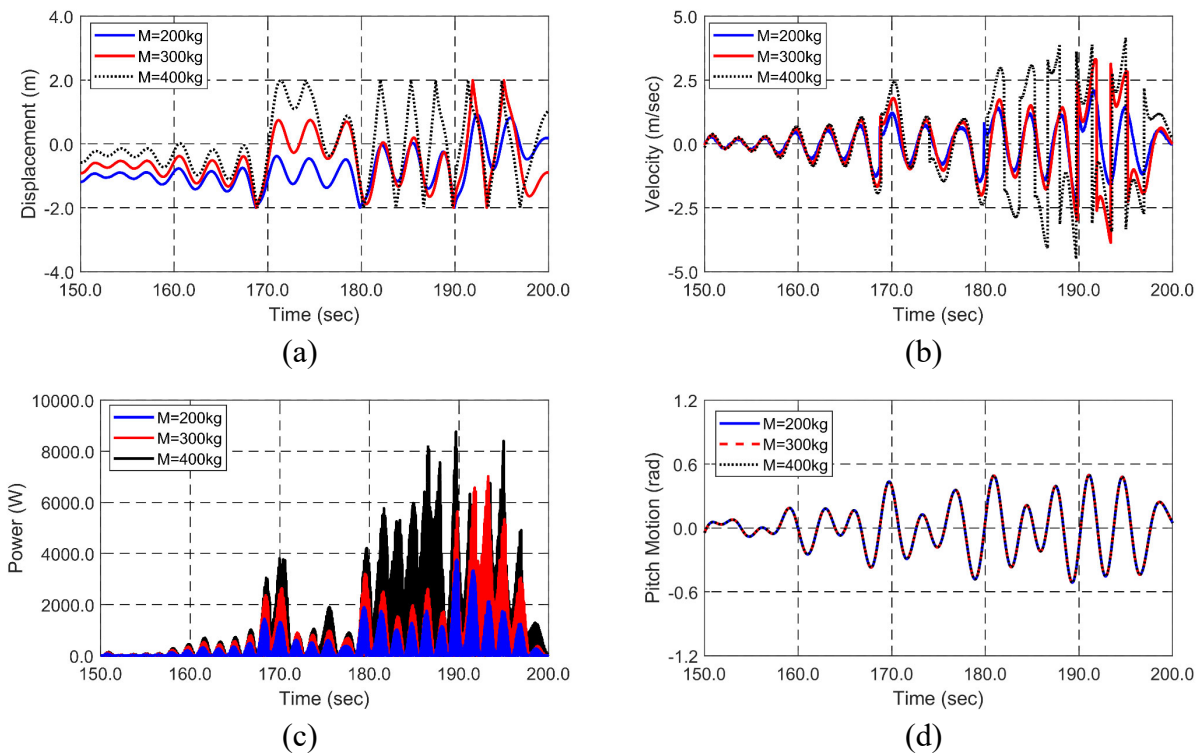
611 Fig. 18. Average output power at different coefficients of restitution and wave conditions.

612

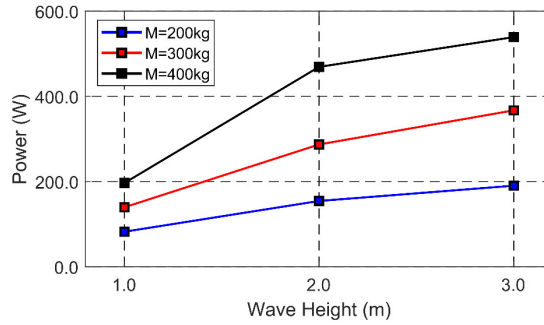
613 5.2.5. Parametric Study 4: Mass of the Magnet Assembly

614 Since gravity is the primary source of power generation, the mass of the magnet assembly
 615 can significantly affect the sliding performance. In this regard, the magnet assembly's mass was
 616 varied in the simulations from 200 kg to 400 kg whereas the previous best parameters, including

617 the coefficient of restitution of 0.8, were fixed. Fig. 19 shows the time histories of displacement
 618 and velocity of the magnet assembly in the sliding direction, output power, and pitch motion of
 619 the outer cylinder at different magnet masses. As shown in Fig. 19(a-c), the heavier the mass of
 620 the magnet assembly, the higher the sliding kinematics and output power. Moreover, significant
 621 variations among the cases are noticeably detected when contacts at both ends take place under the
 622 substantial pitch motion (180-200 sec). The average output power is presented in Fig. 20 for
 623 various magnet masses and sea states, and it also supports the previously mentioned trend. Thus,
 624 the mass of magnet assembly=400 kg was used in the previous simulations. Of course, much larger
 625 magnet mass beyond 400 kg will generate higher power but may cause larger structural impacts at
 626 both ends and also affect the pitch performance through the moment induced by the magnet mass
 627 and time-varying pitch mass moment of inertia. The center rod should also be strong enough to
 628 support a much larger magnet mass without bending.
 629



630 Fig. 19. Time histories of displacement (a) and velocity (b) of the magnet assembly, output
 631 power (c), and pitch motion of outer cylinder (d) at different masses of the magnet assembly
 632 ($H_s=2$ m, $T_p=6$ sec).
 633



634 Fig. 20. Average output power at different masses of the magnet assembly and wave conditions.

635

636 **5.2.6. Output-Power Calculations in Various Random Wave Conditions**

637 The previous parametric studies demonstrate that adjustment of various system parameters
 638 can significantly improve the sliding performance and output power under 3 different wave
 639 conditions, as shown in Figs. 14, 16, 18, and 20. Much more time-domain simulations with
 640 enlarged random wave conditions were additionally performed to check the corresponding output
 641 power by using the parameters selected from the previous sections. The selected values of EMF
 642 magnitude, load resistance, travel length, coefficient of restitution, and magnet mass were 300
 643 T·m, 80 Ω, 4 m, 0.8, and 400 kg, respectively, denoted as Case 1. As summarized in Table 6, we
 644 additionally simulated four more cases (Cases 2-5) to compare whether the chosen parameters can
 645 provide high output power.

646

647 Table 6. Case description.

Case number	EMF magnitude (T·m)	Load resistance (Ω)	Travel length (m)	Coefficient of restitution	Magnet mass (kg)
1	300	80	4	0.8	400
2	200	40	4	0.8	400
3	300	80	6	0.8	400
4	300	80	4	0.38	400
5	300	80	4	0.8	300

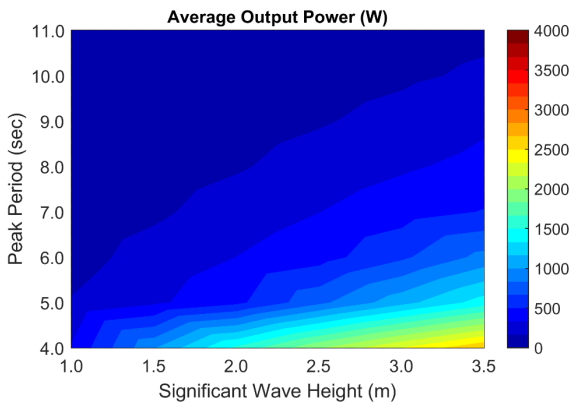
648

649 Fig. 21 shows the average output powers and RMS (root mean square) values of pitch
 650 motion of the outer cylinder at different significant wave heights from 1 m to 3.5 m and peak
 651 periods from 4 sec to 11 sec. Recall that the pitch natural frequency is 1.65 rad/s (3.81 sec). In this
 652 case, large pitch motions usually occur at low peak periods being closer to the pitch natural period,
 653 as shown in Fig. 21(f). Larger wave height also causes larger pitch motion under the identical peak
 654 period. Combining these two facts, it can be expected that the largest average output power occurs
 655 at the bottom-right corner of Fig.21(a-e) i.e. the lowest peak period and the highest wave height
 656 ($H_S=3.5$ m, $T_P=4$ sec). For Case 1, the corresponding average output power under the wave
 657 condition is 2.66 kW. However, even with $H_S=3.5$ m at $T_P=11$ sec (right-top corner), the average
 658 output power is only 0.13 kW. Therefore, the present SR-WEC is particularly designed to be

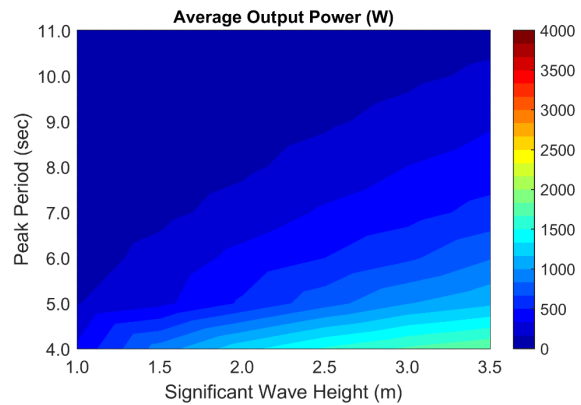
659 efficient at low sea states, which is good since they cover more than 90% of typical annual sea
 660 states. In general, Case 1 shows the relatively high average output power compared with other
 661 cases. There are several environmental conditions where Case 2 has slightly higher output power
 662 than Case 1. Noticeably, in some conditions with the significant pitch motion at high wave height
 663 and low peak period, Case 3, in which travel length is 6 m, generates higher power than Case 1.
 664 For example, the maximum output power is 3.82 kW for Case 3 at $H_S=3.5$ m, $T_P=4$ sec. However,
 665 Case 1 still produces higher annual average power in that the occurrence of such wave condition
 666 is low. These comparisons support the parametric studies' role in finding a good combination of
 667 parameters to improve output power. The metaheuristic optimization algorithms such as the
 668 genetic algorithm and harmony search can be applied to optimize parameters to enhance output
 669 power further, and the results of the present parametric studies can be utilized to define the upper
 670 and lower boundaries of each parameter.

671 We calculated the capture width ratios for Case 1 with the average wave power and
 672 generated output power under random wave excitations [41], which are 9.3% at $H_S = 1.5$ m and T_P
 673 = 4 sec and 3.3% at $H_S = 1$ m and $T_P = 5$ sec.

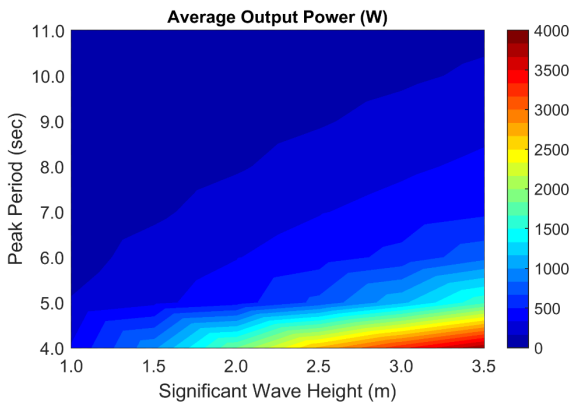
674



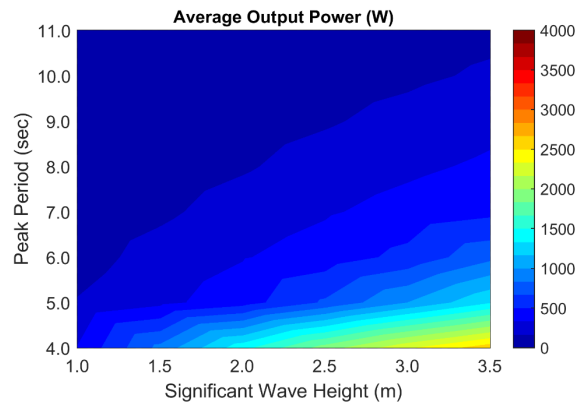
(a) Case 1



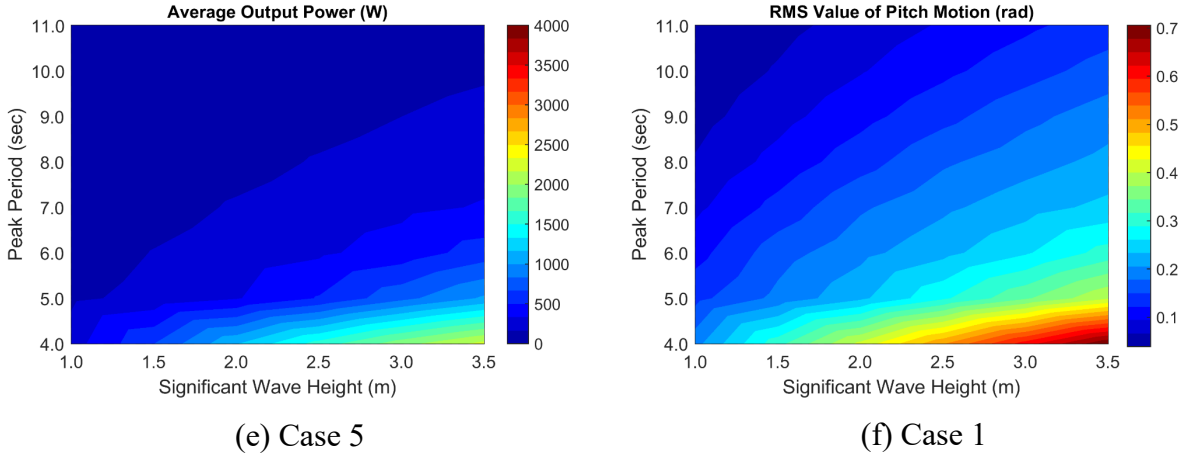
(b) Case 2



(c) Case 3

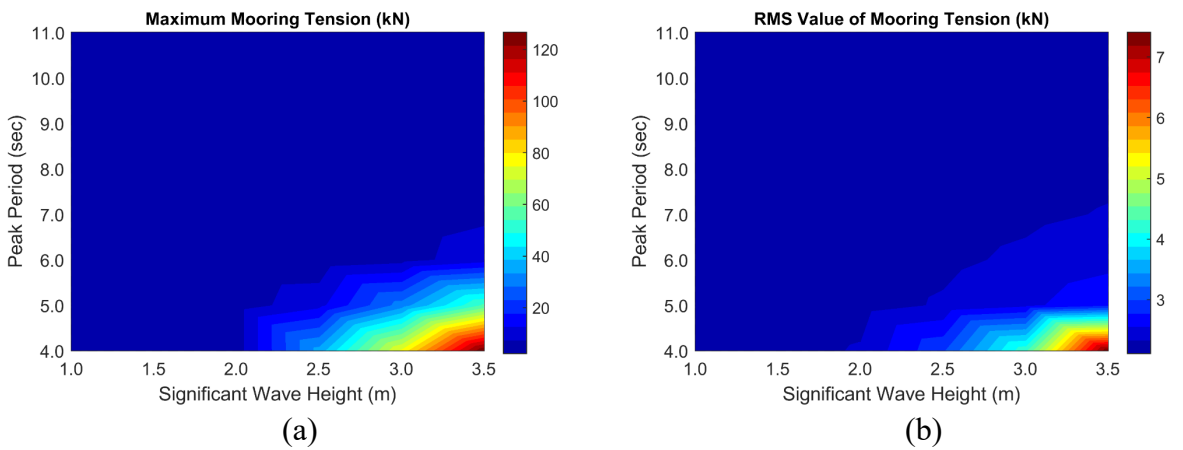


(d) Case 4



675 Fig. 21. Average output powers for Cases 1-5 (a-e) and RMS value of pitch motion of outer
 676 cylinder for Case 1 (f) under different random wave excitations.

677
 678 Finally, we also checked the safety of the mooring line for various sea states, as presented
 679 in Fig. 22. In this case, studlink R4 chain was used, whose minimum breaking load (MBL) is 263.9
 680 kN. When the safety factor of 1.67 is applied, as suggested by the API-mooring-design guideline,
 681 the allowable maximum mooring tension is 158 kN. In the present simulations, the maximum
 682 mooring tension is 126.7 kN in the case of $H_s=3.5$ m and $T_P=4$ sec, for which the largest outer-
 683 cylinder pitch motions and maximum power generation occur. It proves that the given mooring
 684 design is acceptable. In addition, the RMS value of mooring tension is large at high significant
 685 wave height and low peak period, similar to output power and pitch motion.
 686



687 Fig. 22. Maximum mooring tension (a) and RMS value (b) of mooring tension under different
 688 random wave excitations (Case 1).

689
 690 **6. Conclusions**

691 In this study, the performance of SR-WEC was evaluated by the floater-mooring-LEG

692 coupled time-domain simulations. SR-WEC consists of two bodies: the outer cylinder with an
693 armature assembly (body #1) and a magnet assembly (body #2) sliding inside. For SR-WEC, the
694 sliding displacement and velocity are mainly caused by gravity acceleration and the outer
695 cylinder's motions, and they lead to electrical power generation.

696 During the developing stage, the proposed simulation tool was verified step by step by
697 comparisons with laboratory tests. First, the numerically simulated sliding displacement and
698 velocity were compared with those of heave-pitch coupled actuator tests by authors, which shows
699 a good match. Second, the present generator-dynamics numerical solver was verified by the LEG
700 experiment of [10], and the overall comparison is also excellent.

701 After verifying the simulation program, a series of parametric studies were carried out
702 under different random wave conditions to find the appropriate design with increased output
703 power. The selected design parameters were the magnitude of EMF, load resistance, travel length,
704 mass of the magnet assembly, and restitution coefficient of end dampers. The time histories of
705 sliding displacement and velocity of the magnet assembly and output power show that the
706 performance of SR-WEC depends appreciably on those design parameters. After setting all the
707 optimized parameters, the average output power was evaluated under largely extended random-
708 wave conditions. The best performance of the SR-WEC is observed when the wave peak period is
709 close to its natural period with the highest wave height. The resulting peak average output power
710 is 2.66 kW at $H_S=3.5$ m and $T_p=4$ sec. By riding along the surface without much structural
711 resistance, mooring and structural designs can have benefits.

712 The SR-WEC performance can further be improved with semi-active and/or active control
713 systems, such as the movable ring-type masses adjusting pitch natural frequency, latching control,
714 and PTO-force control, which will be the subject of next study.

715

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720

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